

# Canisterized Satellite Dispenser (CSD) As A Standard For Integrating and Dispensing Hosted Payloads on Large Spacecraft and Launch Vehicles

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# What are Canisterized Satellite Dispensers (CSD)

• A rectangular box that dispenses a payload between 1 and 30 Kg spacecraft





# Why is dispensing from a hosted payload valuable

- For the <u>hosted</u> payload it is an excellent launch service:
  - Reliable and frequent opportunities
  - Cost effective
  - Power and telemetry prior to dispensing
- For the <u>hoster</u>:
  - Added revenue for launch service
    - several \$M for 15 kg payload
  - As a co-orbiting spacecraft
    - Observing deployment
    - Increasing system aperture
    - Confusing adversaries

Hoster (ISS)	Hosted payloads (satellites)

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# An example mission: Lunar Water Distribution (LWaDi)

- A 10 kg, 200 watt NASA "CubeSat"
- Looks for water on the moon
- Uses electric propulsion to attain the needed 1.6 km/sec from Geostationary to moon





- Hosted payload must
  - Not present a credible risk to the primary
    - Fail safe
  - Be very easily integrated to many launch streams
  - Need trivial power and telemetry
  - Have a mass well within margin of hoster



#### **Electrical Standard**

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## **Electrical Initiation of Dispensing**





#### **Mechanical Standard**





Table 1: Test Levels			
Qualification	Flight		

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Bench top Separations (1)	200 separations	10 separations	10 separations	
Thermal Vacuum	Temperature: -45°C to +90°C	Temperature: -20°C to +70°C	2	
	Pressure: <10E-5 torr	Pressure: <10E-5 torr		
	Cycles: 27	Cycles: 8	10.000000000000000000000000000000000000	
	9 Separations +90°C: 22V, 28V, 36 V +23°C: 22V, 28V, 36 V -45°C: 22V, 28V, 36V	1 Separation (hot or cold, 22V)	Not Tested	
Strength as Sine Burst (3)	Level: 50g (3U), 40g (6U) Cycles: 5 per axis	Level: 20g Cycles: 5 per axis	Not Tested	
Random Vibration (2,3)	Level: 14.1 Grms	Level: 10.0 Grms		
	Duration: 3 min/axis	Duration: 1 min/axis	Not Tested	
	Payload Mass: Maximum	Payload Mass: Maximum		
Shock (2,3)	See Figure 15 3 impacts per axis	Not Tested	Not Tested	

(1) 1atm, ~23°C.

Test

(2) Full qualification was performed with CSD mounted via –Y face. Contact PSC if planning to mount CSD via any other face.

(3) 3U qualified with 6.1 kg payload. 6U qualified with 9.1 kg payload. Contact PSC if 6U payload is heavier than 9.1 kg.



Figure 15: Shock Levels

EDU



# Advantages of CSD Standard



- 6U CSD has been adopted by four manifested DoD payloads; three NASA interplanetary and many university teams
- 6U CSD is available now
  - Revenue generating activity can begin now
  - Not a new development subject schedule slip, technical anomalies and extensive vetting
- CSD has zero refurbishment cost
- All procedures to integrate operate are ready and vetted by ORS, AFRL, NASA, SpaceX and the DoD



- **Qualified** for use on all US launch vehicles
  - Per Mil-STD-1540 and NASA GEVS
- Preloaded Payload Tabs create a predictable load path to and from the payload. In other words a \$30M payload won't jiggle creating a hang-fire risk.
- Lower tipoff Tabs, roller bearing and linear way minimize tipoff allowing payloads to detumble faster and avoid saturating ADACS
- CSD Constrained Deployables eliminate the payload's need for complex restraint mechanisms.
- Six Mountable Sides increase integration options. Hoster needs to qualify less structure and can fly more revenue generating mass



 Motor Driven Initiator creates a reliable and testable deployment mechanism that automatically resets without consumables. NanoRacks lowers recurring

cost.

# **Technical benefits**

- **Constant force ejection spring** even at the end of stroke the ejector plate is pushing the payload out. Choosing one to four springs allows broad control over dispensing velocity
- Manual override allowing technicians to easily dispense payload

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- Smallest External Volume versus existing designs increases packaging density on launch vehicle.
- No tools required to stow or dispense



6U Payload integration





## **Technical benefits**

- PLANETARY Systems Corporation
- Largest Internal
   Volume versus existing designs accommodates
   11% larger payloads.
   Payloaders can easily add more deployable
   solar panels to make
   their mission better
- Backwards

   Compatible
   Mechanical Interface
   ensures compatibility
   with existing 1, 2, 3 and
   6U Cubesats



The rail kit allows the CSDs compatibility with other standards



The CDS standard has 11% more volume allowing end users more room for solar panels, antennas and thermal treatments



- **Reverse polarity protection** The CSD will properly initiate even if the wired backwards
- Separation Electrical Connector allows communication and charging between payload and launch vehicle. Can support high end / high revenue missions
- State switches Indicate door state; payload occupancy and dispensing velocity
- Electrical redundancy two independent circuits to the motor. Triple redundant commutator
- Conductive outer surface eliminates surface charging







- Safe/Arm Access on Front Door ensures payload access at all times.
- Parametric design of CSDs Commonality allows users easy understanding of electro-mechanical interface: 1/8U, 3U, 6U and 12U
- Fully documented Mechanical and electrical interfaces fully defined for CSD and Payload. See http://www.planetarysystemscorp.com/#! \_\_downloads
- CAD models available on request

C ANISTERIZED SATELLITE DISPENSER (CSD) DATA SHEET				
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	Figure 1: 3U 6U and 12U CSDs. 6U	Shown with Access Panels Removed.		
<ul> <li>Preloaded Payload</li> <li>CSD Constrained D mechanisms.</li> <li>Six Mountable Side adjustion st</li> </ul>	FEATURES AND BENEFITS Table create a predictable load point to and from the payload. heployables eliminate the payload's need for complex restraint is increase integration options and greatly reduce the need for	TABLE OF CONTENTS Mechanical Interface Exercise Interface	4 6 9	
<ul> <li>Motor Driv automatical</li> </ul>				
<ul> <li>Robust Str</li> <li>Separation</li> </ul>	PAYLOAD SPECIFICATIO	IN FOR 30, 60, 120 AND 270		
<ul> <li>Conductive</li> </ul>	This is a standalone specification intended for payload designer	s. Planetary Systems Cornoration does not design or manufacture		
Complete      Completels	payloads but can recommend vendors.	a name a y systems corporation does not catagin or mana active	13	
Manual Do     P.Pod Con	DESCRIPTION			
Full Lengtl	These payloads are designed to be fully contained within a Canisterized Satellite Dispenser (CSD, canister or dispenser)			
vehicle.	during launch. A CSD encapsulates the payload during launch and dispenses it on orbit. CSDs reduce risk to the primary payload and		22	
Largest Int     Safe/Arm /	therefore maximize potential launch opportunity. They also ease restrictions on pavload materials and components. This			
_	specification currently encompasses four payload sizes, 3U, 8U, 12U and 27U		22	
The CSD is com CSD is also com	The payloads incomprate two tabs running the length of the	34 Tru		
	ejection axis. The CSD will grip these tabs, providing a secure, modelable, preloaded junction	Al Engineers on		
2002337 Rev A	moceacie, precaseo junction. The payload may use the CSD to restrain deployables. The allowable contact zones are defined. A payload can be built to this specification without knowledge of		1/22	
	the specific dispense within it will fly. Similarly, dispenser manufactures will be ensured of compatibility with payloads that conform to this specification.	The state of the s		
	Patra	Figure 2: Pavloads		
	Payload (this specification)			
		Revision Release Date Created By Reviewed By		
		- 25-Jul-2012 RH WH		
		Changes from previous revision: Parameters: Changed EL, DCY. Common Requirements: Note 1, changed tab material from 8081-178 br 2078-77 and changed surface finish		
	Figure 1: Payload Deploying from CSD	from chem-film to hard anotize. Added several more notes. 9 Figure 5: Changed Tab filets. Increased payload volume near Tabs. Added missing dimensions. 9 Recommended Tar 45 Antegration: Changed reference		
	CONTENTS Besteine History	<ul> <li>Added Predicting Design Limit Loads.</li> </ul>		
	Parameters 2 Common Beneficiements 2	<ul> <li>Added Separation Electrical Connector Attachment.</li> <li>Added Tips and Considerations.</li> </ul>		
	Electrical Schematic 4	Added Anticipated Improvements.		
	Benefit of Tabs			
	CSD Constrained Deployables			
	Separation Electrical Connector Attachment			
	recommended test and integration			
	CAD Models			
	Anticipated Improvements 12 Authors 12			
	2002367 Rev A 08-Aug-2013	PLoansam Reserve Diserverse Diserverserve Diserverserve 1/12		



## **Thank You**

• Questions?